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Test and Evaluation Report of the Impact Portable Aspirator Model 325M

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The Impact Portable Aspirator, Model 325M, was tested for electromagnetic interference/compatibility in the UH-60A helicopter under the U.S. Army Program for Testing and Evaluation of Equipment for Aeromedical Operations. The tests were conducted using current military and industrial standards and procedures for electromagnetic interference/compatibility and human factors. The Impact Portable Aspirator, Model 325M, was found to be compatible with the U.S. Army MEDEVAC UH-60 Black Hawk.

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Section 1. Executive digest

The Army program for Test and Evaluation of Aeromedical Equipment uses existing military standards (MIL-STD) and collective professional expertise to test and evaluate selected medical equipment proposed for use aboard Army aircraft. Equipment meeting these standards ensures the safety of the crew, patients, and aircraft by eliminating risks due to: (1) Interference by the medical equipment with aircraft systems/subsystems operation, (2) the aircraft system's interference with the operation of the medical equipment, (3) the medical equipment's susceptibility to environmental exposure, or (4) physical and/or functional incompatibility while in use on board selected rotary-wing aircraft. This program tests both developmental and nondevelopmental (off the shelf) medical equipment destined for use aboard Army medical evacuation (MEDEVAC) aircraft.

1.1 TEST OBJECTIVES

1.1.1 To determine if the medical equipment is complete and operational per the manufacturer's operating instructions.

1.1.2 To ensure the electrical safety of the medical equipment.

1.1.3 To ensure the equipment will function as designed throughout the rated battery operation time.

1.1.4 To ensure the safety of the operator, the patient, and the aircrew.

1.1.5 To assess design considerations which could potentially contribute to an operator error.

1.1.6 To determine if the medical equipment can function as designed in a low pressure environment.

1.1.7 To determine the ability of the medical equipment to withstand the vibrational stresses expected in a rotary-wing flight environment without degradation or malfunction.

1.1.8 To determine the ability of the medical equipment to be stored and operated in a high temperature environment.

1.1.9 To determine the ability of the medical equipment to be stored and operated in a low temperature environment.

1.1.10 To determine the ability of the medical equipment to operate satisfactorily for short periods during exposure to high humidity conditions.

1.1.11 To assess the levels of electromagnetic emissions produced by the medical equipment within selected frequency ranges.

1.1.12 To assess the minimum electromagnetic susceptibility levels of the medical equipment within selected frequency ranges.

1.1.13 To assess the physical and/or functional compatibility of the medical equipment while in use on board the aircraft.

1.1.14 To assess the electromagnetic interference (EMI) and electromagnetic compatibility (EMC) characteristics of the medical equipment with the host aircraft and its installed systems.

1.2 TESTING AUTHORITY

Research and Technology Work Unit Summary, dated 5 October 1989. Project number 3M463807D836, titled, Army Program for Testing and Evaluation of Equipment for Aeromedical Operations.

1.3 SCOPE

1.3.1 This test was conducted at the United States Army Aeromedical Research Laboratory (USAARL), Cairns Army Airfield (CAAF), and designated test flight areas in and around Fort Rucker, Alabama.

1.3.2 The USAARL UH-60A aircraft, serial number 88-26069, with subsystems delineated in paragraph 3.2.2, was configured with the Impact Portable Aspirator*, model 325M and used as the test aircraft for the in-flight evaluation. The in-flight evaluation required 2.0 flight hours.

1.3.3 Prior to flight testing, the following tests were accomplished: Acceptance inspection, equipment training, electromagnetic compatibility, human factors and safety, environmental compatibility, and in-flight compatibility.

1.3.4 An airworthiness release (AWR) was received from the U.S. Army Aviation Systems Command (AVSCOM) prior to the in-flight testing of the Impact Portable Aspirator.

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* See list of manufacturers

1.4 MATERIAL DESCRIPTION

The Impact Portable Aspirator, Model 325M, features high vacuum and high free airflow to remove blood, vomitus, secretions, and debris from the airway. This is required for patients incapable of clearing their own secretions. The unit's main components include an electric motor and cylinder assembly. Suction is created when the piston is moved in either direction and the vacuum is transferred through a collection bottle. The unit is self-contained, portable, and powered by an internal 12 volt battery, 12 volt dc power source, or 120 Vac, 60 Hz power source. The internal battery can be recharged from a 120 Vac, 60 Hz power source.

1.5 SUMMARY

1.5.1 Laboratory testing

1.5.1.1 Battery Life Evaluation: The Impact Aspirator operated an average of 1 hour and 4 minutes from a fully charged battery. This exceeds the manufacturer's specification of 1 hour operation.

1.5.1.2 Electrical Safety Evaluation: No unsafe qualities were found in the Impact Model 325M. The limits for currents and resistances were in accordance with (IAW) the limits specified in TB-38-750-2, April 1987 and National Fire Prevention Association (NFPA) standards.

1.5.1.3 Human Factors Evaluation: The Impact Model 325M was found to be satisfactory in all categories of the evaluation.

1.5.1.4 Environmental Tests: The Impact Model 325M can be expected to perform in a variety of environmental conditions. Its performance was found to be satisfactory in all stages of the environmental testing. The requirements for environmental tests are established in MIL-STD-810D, Methods 500.2 (altitude), 514.3 (vibration), 501.2 (high temperature), 502.2 (low temperature), and 507.2 (humidity).

1.5.1.5 Radiated Emissions Tests (RE02): The Impact Model 325M may be unsatisfactory for use in certain EMI sensitive environments. A broadband (BB) radiated emission was detected in the test frequency range that exceeded the test limits. Emission limits are set forth in MIL-STD-461A, Notice 4.

1.5.1.6 Radiated Susceptibility Test (RS03): The Impact Model 325M was not found to be susceptible to radio frequency interference in the testing range and magnitude.

1.5.2 In-flight testing

1.5.2.1 During the in-flight human factors evaluation, the Impact Model 325M was found to be satisfactory in all categories of the evaluation criteria. The battery provided power for 45 minutes with the light on and the heater set for 90°F (58°F ambient temperature). If ac power is not available, the battery power may not be sufficent for a long transport in cold weather. The unit functioned properly in the aircraft with ac power.

1.5.2.2 The aircraft and its subsystems were not adversely affected by the operation of the Impact Model 325M in any of the prescribed flight test modes.

1.5.2.3 The Impact Model 325M was not affected by the aircraft and its subsystems during the in-flight testing.

1.6 CONCLUSION

Based on the results of laboratory and in-flight testing, the Impact Portable Aspirator, Model 325M was found to be compatible with U.S. Army MEDEVAC UH-60A Black Hawk with the subsystems listed in paragraph 3.2.2.

Section 2. Subtests

2.1 INITIAL INSPECTION

2.1.1 Objective

To determine if the Impact Model 325M is complete and operational for testing per the manufacturer's operating instructions.

2.1.2 Criteria

2.1.2.1 The physical inventory is conducted solely for investigation and documentation.

2.1.2.2 The Impact Model 325M will display consistent and accurate performance as an acceptable performance test.

2.1.3 Test procedure

2.1.3.1 A complete physical inventory of the Impact Model 325M was completed per the manufacturer's equipment list.

2.1.3.2 An operational validation test of the Impact Model 325M was conducted per the manufacturer's operating instructions by USAARL's medical maintenance personnel.

2.1.4 Test findings

2.1.4.1 The Impact Model 325M was inventoried and found to be complete.

2.1.4.2 The Impact Model 325M operated as prescribed in the manufacturer's operating manual. Criteria met.

2.2 BATTERY LIFE EVALUATION (Laboratory)

2.2.1 Objective

To ensure the equipment will function as designed throughout the rated battery operation time.

2.2.2 Criterion

Verify manufacturer's specified full power internal battery life expectancy of 1 hour operation.

2.2.3 Test procedure

2.2.3.1 Charging and operation cycles were conducted in ambient room conditions.

2.2.4 Test findings

The unit operated an average of 1 hour and 4 minutes from a fully charged battery. This exceeds the manufacturer's specified battery life. Criterion met.

2.3 ELECTRICAL SAFETY EVALUATION

2.3.1 Objective

To ensure the electrical safety, by evaluation of case-to-ground resistance and case-to-ground current leakage, of the Impact Model 325M.

2.3.2 Criterion

The Impact Model 325M shall meet the standards established in TB-38-750-2 and NFPA 99 for electrical safety of medical equipment.

2.3.3 Test procedure

Performance in the electrical safety evaluation are made, with a Neurodyne-Dempsey model 431F electrical safety analyzer*, IAW the procedures described in Technical Bulletin (TB) Number 38-750-2. Case-to-ground resistance and various case-to-ground leakage currents were measured. Leakage currents were measured using a 10 by 20 centimeter (cm) aluminum foil sheet taped flush to the equipment case. Checks were made for safety concerns such as case integrity, breaks in power cord insulation, and connectors.

2.3.4 Test findings

Since there are no patient or power leads required for operation of the unit from battery power and the case of the unit is made from non-conductive plastic, the case-to-ground resistance and leakage currents could not be measured. No unsafe qualities were found in the case integrity, charging adapter, or connectors. Criterion met.

2.4 HUMAN FACTORS EVALUATION (Laboratory)

2.4.1 Objectives

2.4.1.1 To assure the safety of the operator, the potential patient, and the aircrew.

2.4.1.2 To assess the design considerations which could potentially contribute to an operator error.

2.4.2 Criterion

The Impact Model 325M must be rated satisfactory in all major categories of the evaluation. These include visual displays, controls, maintainability, conductors, fasteners, test points, test equipment, fuses and circuit breakers, labels and coding, and safety.

2.4.3 Test procedure

2.4.3.1 The evaluation was conducted in a laboratory under fluorescent lighting and ambient room conditions.

2.4.3.2 The Impact Model 325M was operated according to prescribed instructions through its full range of functions.

2.4.4 Test finding

The Impact Model 325M was found to be satisfactory in all of the evaluation criteria. Criterion met.

2.5 ALTITUDE (LOW PRESSURE) TEST [IAW MIL-STD-810D, METHOD 500.2]

2.5.1 Objective

To determine if the Impact Model 325M can function as designed in a low pressure environment.

2.5.2 Criterion

The Impact Model 325M will perform as designed while exposed to an altitude equivalency of 15,000 feet above sea level.

2.5.3 Test procedure

2.5.3.1 A pretest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.5.3.2 The altitude test was performed in a Tenney Engineering model 64S altitude chamber*. This test is based on MIL-STD-810D, Method 500.2. The Impact Model 325M was operated on the floor of the chamber. Chamber pressure was decreased to 420 mmHg (15,000 ft equivalent altitude) over a 15-minute period, held constant for 60 minutes, then raised, at 1500 fpm, to ambient conditions (760 mmHg) over a 10-minute period. There are no provisions for the control of temperature or humidity inside this chamber.

2.5.3.3 A posttest performance check was conducted to ensure proper operation of the Impact Model 325M after the exposure to low pressure.

2.5.4 Test findings

2.5.4.1 The pretest performance check met criterion 2.1.2.2.

2.5.4.2 No failures in the performance of the Impact Model 325M were noted before, during, or after the altitude test. Criterion met.

2.5.4.3 The posttest performance check met criterion 2.1.2.2.

2.6 VIBRATION TEST [IAW MIL-STD-810D, METHOD 514.3]

2.6.1 Objective

To determine the ability of the Impact Model 325M to withstand the vibrational stresses expected in a rotary-wing environment without degradation or malfunction.

2.6.2 Criterion

The Impact Model 325M will remain operational and be able to display consistent and accurate performance while exposed to vibrational stresses.

2.6.3 Test procedure

2.6.3.1 A pretest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.6.3.2 The vibration test was performed using an Unholtz-Dickey model TA115-40/CSTA vibration test system*. It is a single-axis system with an electromagnetic driver unit. The test consisted of sinusoidal vibrations superimposed on random vibrations over a frequency range of 500 Hz, as shown below. The reference spectrum breakpoints are from MIL-STD-810D, Method 514.3; reference spectrum levels are based on field performance with a conservatism factor of 1.5. Independent tests were conducted in the X, Y, and Z axes.

Z-axis
duration: 60 minutes
broadband intensity: 0.4506 G_{rms}
random vibration: initial slope : 99.00 dB/oct
5 Hz level: 0.00006210 $G_{sqr/Hz}$
100 Hz level: 0.0006210 $G_{sqr/Hz}$
300 Hz level: 0.0006210 $G_{sqr/Hz}$
500 Hz level: 0.00006210 $G_{sqr/Hz}$
final slope: -99.00 dB/oct
sinusoidal vibration: .5450 G_{pk} at 11.25 Hz
.1690 G_{pk} at 22.50 Hz

.1200 G_{pk} at 33.75 Hz
.0310 G_{pk} at 45.00 Hz
.0530 G_{pk} at 56.25 Hz

X and Y axes
duration: 60 minutes each
broadband intensity: 0.3099 G_{rms}
random vibration: initial slope: 99.00 dB/oct
5 Hz level: 0.00002920 $G_{sqr/Hz}$
100 Hz level: 0.0002920 $G_{sqr/Hz}$
300 Hz level: 0.0002920 $G_{sqr/Hz}$
500 Hz level: 0.00002920 $G_{sqr/Hz}$
final slope: -99.00 dB/oct
sinusoidal vibration: .3200 G_{pk} at 11.25 Hz
.0670 G_{pk} at 22.50 Hz
.0950 G_{pk} at 33.75 Hz
.0350 G_{pk} at 45.00 Hz
.0770 G_{pk} at 56.25 Hz

The Impact Model 325M was strapped to the vibration table fixture, and its performance was evaluated before, during, and after exposure to vibration.

2.6.3.3 A posttest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.6.4 Test findings

2.6.4.1 The pretest performance check met criterion 2.1.2.2.

2.6.4.2 No failures in the performance of the Impact Model 325M occurred before, during, or after exposure to vibration. Criterion met.

2.6.4.3 The posttest performance check met criterion 2.1.2.2.

2.7 HIGH TEMPERATURE TEST [IAW MIL-STD-810D, METHOD 501.2]

2.7.1 Objective

To determine the ability of the Impact Model 325M to be stored and operated in a high temperature environment.

2.7.2 Criteria

2.7.2.1 The Impact Model 325M will demonstrate consistent and accurate operation during the high temperature operation check.

2.7.2.2 The Impact Model 325M will demonstrate consistent and accurate operation after the high temperature storage cycle.

2.7.3 Test procedure

2.7.3.1 A pretest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.7.3.2 The high temperature test was conducted in a Tenney Engineering model ZWUL-10107D walk-in controlled environment chamber*. This test is based on MIL-STD-810D, Method 501.2. For the high temperature operation test, the Impact Model 325M was turned on and placed on the floor of the environmental chamber. The chamber temperature was raised to 49°C and the humidity was stabilized at a maximum of 20 percent relative humidity (RH) within 15 minutes. The environmental control system is capable of regulating temperature within ± 2°C and humidity within ± 5 percent RH. Temperature and humidity were held constant for 2 hours. At 30-minute intervals, the chamber door was opened briefly to minimize the change in chamber conditions during performance checks. After the operational test, the Impact Model 325M was allowed to return to ambient conditions over a 30-minute period.

2.7.3.3 A posttest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.7.3.4 The Impact Model 325M was stored (not operated) at temperatures of 63°C for 1 hour, 71°C for 4 hours, then again at 63°C for 1 hour. The chamber and Impact Model 325M then were returned to ambient conditions over a 30-minute period.

2.7.3.5 A poststorage performance check was conducted to ensure proper performance of the Impact Model 325M.

2.7.4 Test findings

2.7.4.1 The pretest performance check met criterion 2.1.2.2.

2.7.4.2 No operational failures occurred during the high temperature test. Criterion met.

2.7.4.3 The posttest performance check met criterion 2.1.2.2.

2.7.4.4 The Impact Model 325M functioned properly after the high temperature storage test. Criterion met.

2.8 LOW TEMPERATURE TEST [IAW MIL-STD-810D, METHOD 502.2]

2.8.1 Objective

To determine the ability of the Impact Model 325M to be stored and operated in a low temperature environment.

2.8.2 Criteria

2.8.2.1 The Impact Model 325M will demonstrate consistent and accurate operation during the low temperature operation check.

2.8.2.2 The Impact Model 325M will demonstrate consistent and accurate operation after the low temperature storage cycle.

2.8.3 Test procedure

2.8.3.1 A pretest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.8.3.2 The Impact Model 325M was placed on the floor of the environmental chamber and the temperature was lowered to -25°C within 25 minutes. The environmental control system is capable of regulating temperature within 2°C. Humidity cannot be controlled in the chamber at freezing temperatures. The temperature was held constant for 2 hours. The chamber door was opened briefly every 30 minutes to minimize the change in chamber conditions, and a performance check was conducted. The chamber temperature then was raised to ambient temperature within a 30-minute period.

2.8.3.3 A posttest performance check was conducted to ensure proper operation of the Impact Model 325M.

2.8.3.4 The Impact Model 325M was "stored" in a nonoperational mode. The Impact Model 325M was placed on the floor of the environmental test chamber and the temperature was lowered to -46°C for 6 hours. The chamber then was raised to ambient temperature over a 30-minute period.

2.8.3.5 A poststorage performance check was conducted to ensure proper operation of the Impact Model 325M.

2.8.4 Test findings

2.8.4.1 The pretest performance check met criterion 2.1.2.2.

2.8.4.2 No operational failures occurred during the low temperature test. Criterion met.

2.8.4.3 The posttest performance check met criterion 2.1.2.2.

2.8.4.4 The Impact Model 325M functioned properly after the low temperature storage test. Criterion met.

2.9 HUMIDITY TEST [IAW MIL-STD-810D, METHOD 507.2]

2.9.1 Objective

To determine the ability of the Impact Model 325M to operate for short periods of time during exposure to highly humid conditions.

2.9.2 Criterion

The Impact Model 325M will demonstrate consistent and accurate operation while exposed to a high humidity environment.

2.9.3 Test procedure

2.9.3.1 A pretest performance check was conducted to ensure the proper operation of the Impact Model 325M.

2.9.3.2 The humidity test was conducted in a Tenney Engineering model ZWUL-10107D walk-in controlled environment chamber*. This test is based on MIL-STD-810D, Method 507.2. For the humidity test, the Impact Model 325M was placed in operation on the floor of the environmental chamber. The chamber temperature was raised to a temperature of 30°C and a relative humidity of 95 percent within 25 minutes. Temperature and relative humidity were maintained for 4 hours. The environmental control system is capable of regulating temperature within ± 2°C and humidity within ± 5 percent RH. At 45-minute intervals the performance of the Impact Model 325M was checked. The chamber door was opened briefly to minimize the change in chamber conditions. The chamber and the Impact Model 325M were returned to ambient conditions before the posttest performance validation check was conducted.

2.9.3.3 A posttest performance check was conducted to ensure the proper operation of the Impact Model 325M.

2.9.4 Test findings

2.9.4.1 The pretest performance check met criterion 2.1.2.2.

2.9.4.2 No failures were noted in the Impact Model 325M performance checks conducted during the exposure to the high humidity environment. Criterion met.

2.9.4.3 The posttest performance check met criterion 2.1.2.2.

2.10 ELECTROMAGNETIC CHARACTERISTICS TEST [IAW MIL-STD-461A, Notice 4, AND MIL-STD-462, Notice 3]

2.10.1 Objectives

2.10.1.1 To assess the maximum levels of radiated electromagnetic emissions produced by the Impact Model 325M in the 14 kHz to 12.4 GHz frequency range.

2.10.1.2 To assess the tolerances of radiated electromagnetic susceptibility of the Impact Model 325M within the 10 kHz to 10 GHz electric field.

2.10.2 Criteria

2.10.2.1 The Impact Model 325M will not produce emissions in excess of the limits set forth in MIL-STD-461C, paragraph 6.13.

2.10.2.2 The Impact Model 325M will not malfunction when it is subjected to radiated emissions as specified in MIL-STD-461C, paragraph 6.20.

2.10.3 Test procedure

2.10.3.1 The radiated emissions test was performed according to MIL-STD-462, Notice 3, Method RE02. The Impact Model 325M was positioned on a wooden test stand inside the EMI chamber, 1 meter away from the receiving antennas. The antennas were mounted for both vertical and horizontal polarities and connected to EMI receivers. While the Impact Model 325M was operating, the frequency spectrum (14 kHz to 12.4 GHz) was scanned for emissions. The Impact Model 325M was operated with battery power.

2.10.3.2 The radiated susceptibility test was performed according to MIL-STD-462, Notice 3, Method RS03. The Impact Model 325M was positioned on a wooden test stand inside the EMI chamber 1 meter away from the transmitting antennas. The antennas were mounted for both vertical and horizontal polarities and connected to radio frequency (RF) transmitters. While the Impact Model 325M was operating, it was monitored for faulty operation during exposures to a field of 20 V/m from 10 kHz to 10 GHz. The Impact Model 325M was operated with battery power.

2.10.4 Test findings

2.10.4.1 During the radiated emissions test, a broadband emission exceeding specification limits of MIL-STD-461C, Notice 4, was detected. The emission was detected at 10.75 MHz and exceeded specification by 5.5 dB. Criterion partially met.

2.10.4.2 The Impact Model 325M was not found to be susceptible to radio frequency interference in the testing range and magnitude. Criterion met.

2.11 IN-FLIGHT HUMAN FACTORS EVALUATION

2.11.1 Objective

To assess the physical and/or functional compatibility of the Impact Model 325M while in use onboard the aircraft.

2.11.2 Criterion

The flight surgeon will be able to operate the Impact Model 325M without physical or functional restrictions aboard the aircraft. Major areas of concern include: Proper operation, visual displays, controls, maintainability, conductors, fasteners, test points, test equipment, fuses and circuit breakers, labels and coding, and safety.

2.11.3 Test procedure

2.11.3.1 A human factors evaluation was performed IAW MIL-STD-1472D, AAMI Human factors engineering guidelines, and UL-544 to ensure the compatibility of the Impact Model 325M and the in-flight environment. The flight surgeon conducted the test wearing a flight suit, flight gloves, and an SPH-4B flight helmet. An evaluation of the compatibility with the nuclear, biological, and chemical (NBC) protective equipment was not conducted. Due to restrictions of the AWR, testing was conducted during daylight hours only.

2.11.3.2 The Impact Model 325M was placed on a seat in the aircraft and secured with straps. The Impact Model 325M was tested using battery power in all flight scenarios required by the In-flight Test Operations Procedures (ITOP) (refer to section 3.2).

2.11.4 Test findings

During the in-flight human factors evaluation, the Impact Model 325M was found to be satisfactory in all categories of the evaluation criteria. Criterion met.

2.12 IN-FLIGHT EMI/EMC CHARACTERISTICS

2.12.1 Objective

To assess the EMI/EMC characteristics of the Impact Model 325M with the host aircraft and its installed systems.

2.12.2 Criteria

2.12.2.1 The Impact Model 325M will not radiate EMI to disrupt or interfere with other equipment or systems aboard the aircraft.

2.12.2.2 The aircraft will not radiate EMI to disrupt or interfere with the Impact Model 325M's operation.

2.12.3 Test procedure

A qualitative EMI/EMC assessment was performed with both the Impact Model 325M and the aircraft operating as source and victim. The Impact Model 325M and applicable aircraft instruments and systems were monitored for unusual operation, readings, surges, or power anomalies for each checklist item.

2.12.4 Test findings

2.12.4.1 There were no adverse instances of EMI/EMC noted with the Impact Model 325M acting as either the source or victim. Criterion met.

2.12.4.2 There were no adverse instances of EMI/EMC noted with the aircraft acting as either the source or victim. Criterion met.

Section 3. Supporting documentation

3.1 DETAILED TEST INFORMATION

3.1.1 General information

3.1.1.1 Impact Model 325M testing is not considered a major action significantly affecting the quality of the human environment and, therefore, qualifies for categorical exclusion A-28, appendix A, AR 200-1.

3.1.1.2 A safety pilot will be designated for each flight. Flight operations will be conducted IAW the aircraft operator's manual, appropriate aircrew training manuals, and test item technical data.

3.1.2 Material description

3.1.2.1 The Impact Portable Aspirator, Model 325M, features high vacuum and high free airflow to remove blood, vomitus, secretions, and debris from the airway. This is required for patients incapable of clearing their own secretions. The unit's main components include an electric motor and cylinder assembly. Suction is created when the piston is moved in either direction and the vacuum is transferred through a collection bottle. The unit is self-contained, portable, and powered by an internal 12 volt battery, 12 volt dc power source, or 120 Vac, 60 Hz power source. The internal battery can be recharged from a 120 Vac, 60 Hz power source.

3.1.2.2 Dimensions: 13.5 x 10 x 6.125 in (34 x 25 x 15 cm).

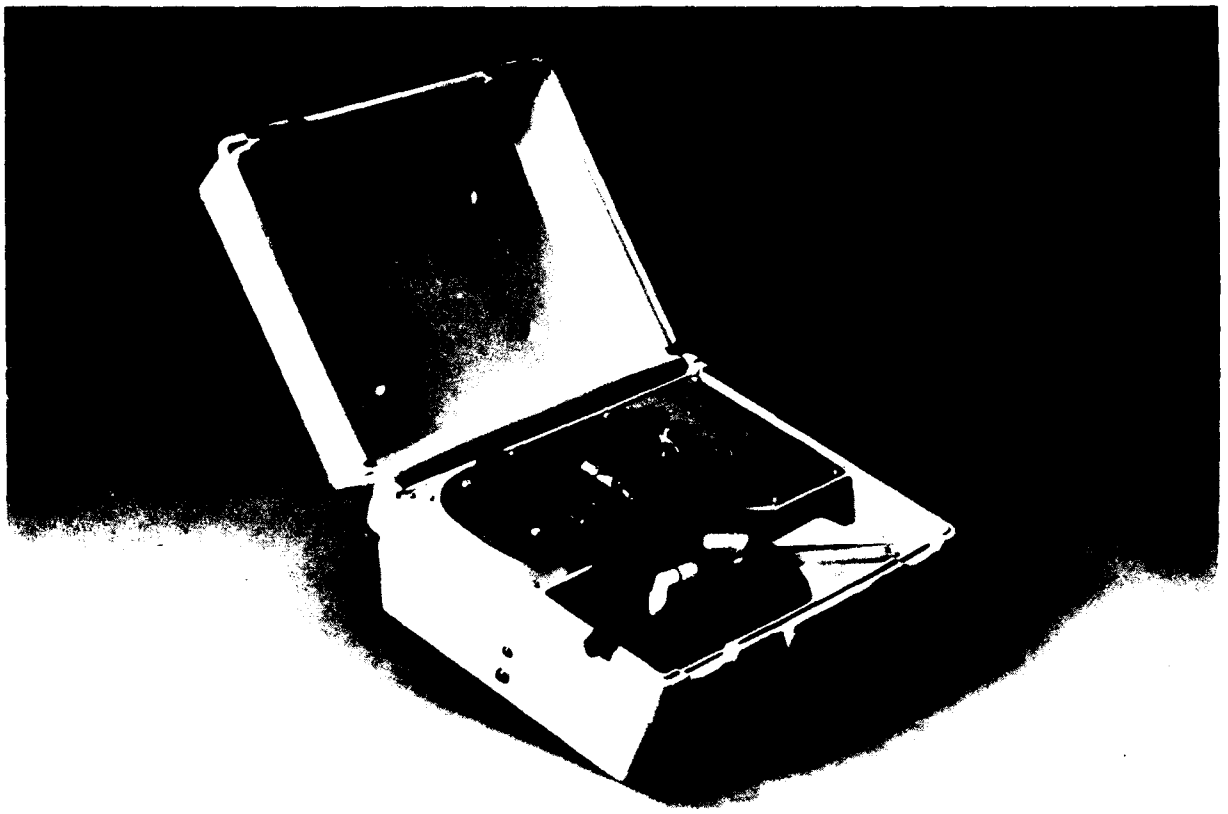
3.1.2.3 Weight: 13 lb (5.9 kg).

3.1.2.4 Standard accessories: collection bottle (680 cc), suction catheter adapter, battery charger.

3.1.2.5 Power requirements: 12 Vdc, or 120 Vac, 60 Hz.

3.2 TEST DATA

3.2.1 Photographic description



3.2.2 Aircraft equipment list

Item No.	Nomenclature
1	Receiver radio -- R-1496A/ARN-89 (automatic direction finder)
2	Displacement gyro -- CN-1314/A
3	Gyro directional -- CN-998/ASN-43
4	Signal data converter -- CV-3338/ASN-128
5	Receiver -- R-2139/ARN-123 (VOR/LOC/MB/GS)
6	Command instrument system processor -- 70600-01038-101
7	SAS amplifier -- 70901-02908-104 (flight control stability augmentation system)
8	Rate gyro -- TRU-2A/A
9	Amplifier, impedance -- AM-4859A/ARN-89
10	Cargo hook -- FE-7590-145
11	Receiver, radar -- RT-1193/ASN-128 (doppler navigation receiver)
12	Barometric altimeter -- AAU-31/A-1
13	Barometric altimeter -- AAU-32A
14	Receiver/transmitter -- RT-1300/ARC-186 (VHF-AM and/or FM radio)
15	UHF-AM radio set -- RT-1518/ARC-164
16	Interphone control -- C6533/ARC (aircraft intercom control)
17	Receiver/transmitter -- RT-1115D/APN-209 (radar altimeter)
18	Indicator altimeter -- ID-1917C/APN-209 (radar altimeter)
19	Control radio set -- C-7392A/ARN-89 (automatic direction finder)
20	Comparator signal data -- CM-482/ARC-186 (comparator for ARC-186)
21	Receiver/transmitter -- RT-1296A/APX-100 (transponder with IFF)
22	Computer display unit -- CP-1252/ASN-128 (doppler navigation system)
23	Compass set controller -- C-8021E/ASN75
24	Magnetic compass - standby -- MS-17983-4

3.2.3 In-flight test data card

DATA CARD FORMAT

GUIDELINE FOR DATA COLLECTION

IN-FLIGHT SUITABILITY TEST OF MEDICAL ITEMS

1. Installation/removal.	Suitable Yes	Suitable No	Comments
a. Weight and balance (DD Form 365-4, Clearance Form F).	X		
b. Space/area allocation.			
(1) Operational requirements.	X		
(2) Storage requirements.	X		
c. Interface connections (safe, positive, secure).	X		
d. Installation/removal (expedient/easily achieved).	X		
e. Mounting/final configuration (functional/stable).	X		

2. Operations and performance.	Suitable Yes	Suitable No	Comments
a. Manufacturer's operating instruction.	X		
b. Medical item operation before aircraft run-up.	X		
c. System interface during aircraft engine run-up and medical item operation (EMI switchology checklist).	X		
(1) Aircraft voltage output.	X		

	Suitable Yes	Suitable No	Comments
(2) Flight control function (UH-60).	X		
(3) Stabilator function (UH-60).	X		
(4) Radio communication vs. medical item operation.			
(a) FM	X		
(b) UHF	X		
(c) VHF	X		
(5) Navigation equipment vs. medical item operation.			
(a) Transponder	X		
(b) ADF	X		
(c) VOR	X		
(d) Doppler	X		
(6) Radar altimeter operation vs. medical item operation.	X		
d. System interface during aircraft hover and medical item operation (EMI switchology checklist).			
(1) Voltage output.	NA		
(2) Radio communication vs. medical item operation.			
(a) FM	X		
(b) UHF	X		
(c) VHF	X		

	Suitable Yes	No	Comments
(3) Navigation equipment operation vs. medical item operation.			
(a) Transponder	X		
(b) ADF	X		
(c) VOR	X		
(d) Doppler	X		
e. Flight mission profile vs. medical item operation (EMI switchology checklist).			
(1) Straight and level (1000 ft MSL for 20 minutes).			
(a) Compatibility of flight mode and medical item operation.	X		
(b) Radio communication vs. medical item operation.			
a. FM	X		
b. UHF	X		
c. VHF	X		
(2) NOE (20 minutes). compatibility of flight mode and medical item operation.	X		
(3) FM homing (10 minutes).	X		
(4) Doppler navigation vs. medical item operation.			
(a) Initialize function.	X		
(b) Fix function.	X		
(c) Update function.	X		

	Suitable Yes	Suitable No	Comments
(5) VOR navigation 7000 ft MSL for 20 minutes) vs. medical item operation.	X		
(6) ILS approach vs. medical item operation.	X		
f. Medical item operation after engine shutdown (external power source).	X		
g. Restrictions to the medical item's use (i.e., electrical connectors).	X		
h. Deviations from the laboratory test results.			
(1) Electrical/ electronic.			None
(2) Mechanical environment.			None
(3) Human factors (user interface, controls, markings, lighting, egress).			None
(4) Safety.			None

3. Deviations from the in-flight test protocol.

a. The VOR navigation portion of the in-flight test conducted at 2000 feet MSL due to air traffic control clearance.

3.2.4 EMI switchology checklist

EMI SWITCHOLOGY CHECKLIST UH-60 AIRCRAFT

IN-FLIGHT SUITABILITY OF MEDICAL ITEMS

ENG INSTRUMENTS/CDU	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Fuel quantity	X			
Fuel indicator test	X			
XMSN oil temperature	X			
XMSN oil pressure	X			
#1 engine oil temperature	X			
#2 engine oil temperature	X			
#1 engine oil pressure	X			
#2 engine oil pressure	X			
#1 TGT	X			
#2 TGT	X			
#1 Ng speed	X			
#2 Ng speed	X			
CDU digits on/off	X			
CDU instruments dim	X			

ENG INSTRUMENTS/PLT PDU	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
#1 engine RPM	X			
#2 engine RPM	X			
Rotor RPM	X			
#1 torque	X			
#2 torque	X			

ENG INSTRUMENTS/COPLT PDU	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
#1 engine RPM	X			
#2 engine RPM	X			
Rotor RPM	X			
#1 torque	X			
#2 torque	X			

ENG CONTROLS	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
#1 overspeed	X			
#2 overspeed	X			
RPM switch	X			
#1 engine anti-ice	X			
#2 engine anti-ice	X			
#1 inlet anti-ice	X			
#2 inlet anti-ice	X			

RADIO EQUIPMENT	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
ICS, C-6533 ARC	X			
VHF-FM, ARC-186/115	X			
VHF-AM, ARC-186/115	X			
UHF-AM, ARC-164(V)	X			
Crypto, KY-28	Not installed			
Radio retransmissions PLN	Not installed			
Transponder, APX-100(V)	X			
KIT-1A/TSEC IFF computer	Not keyed with code			

MISSION EQUIPMENT	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
RWR, APR-39(V)	Not installed			
IR CM, ALQ-144	Not installed			
Chaff dispenser, M-130	Not installed			
Cargo hook system	X			

HYDRAULIC CONTROL SYSTEM	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Backup hydraulic pump	X			
Servo off 1st stage/PLT	X			
Servo off 2nd stage/PLT	X			
Servo off 1st stage/COPLT	X			
Servo off 2nd stage/COPLT	X			
Hydraulic leak test	X			
Tail servo	X			
Boost servos	X			

FUEL SYSTEM	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Fuel pump switch	X			
Fuel boost pump #1	X			
Fuel boost pump #2	X			
Fuel cont panel ESSS	X			

WARNING SYSTEM	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Low rotor RPM	X			
Master caution	X			
Caution advisory	X			
Fire warning	X			
AFCS	X			
Stabilator	X			
#1 engine out	X			
#2 engine out	X			

NAVIGATION INSTRUMENTS	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
ADF	X			
Magnetic compass	X			
CONUS NAV, ARN-123	X			
Doppler, ASN-128	X			
Gyro mag compass (PLT)	X			
Gyro mag compass (COPLT)	X			
Compass cont panel, ASN-75	X			
HSI	X			

FLIGHT INSTRUMENTS	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Radar altimeter	X			
Stabilator pos indicator	X			
VSI	X			
CIS mode select	X			
SAS 1	X			
SAS 2	X			
FPS	X			
Trim	X			
Go-around enable	X			
Cyclic trim release	X			
Cyclic stick trim	X			
ALR encoder	X			

FLIGHT INSTRUMENTS (CONT)	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
HSI/VSI mode select (PLT)				
DPLR	X			
VOR/ILS	X			
BACK CRS	X			
FM HOME	X			
TURN RATE	X			
CRS HDG	X			
VERT GYRO	X			
BRG 2	X			
HSI/VSI Mode Select (COPLT)				
DPLR	X			
VOR/ILS	X			
BACK CRS	X			
FM HOME	X			
TURN RATE	X			
CRS HDG	X			
VERT GYRO	X			
BRG 2	X			

MISCELLANEOUS EQUIPMENT	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Blade deice	Not tested			Ambient temperature was out of test limits.
Windshield anti-ice	X			
Pitot heat	X			
Vent blower	X			
Windshield wiper	X			
Heater	X			
APU	X			
Generator #1	X			
Generator #2	X			
Generator APU	X			
Air source heat start	X			
Tail wheel lock	X			
Gyro erect	X			

LIGHTING	No EMI Affect	EMI Affected Gnd	EMI Affected Flt	Explanation
Cockpit utility	X			
Cockpit flood	X			
Cabin dome	X			
Search light	X			
Search light control	X			
Landing light	X			
Flt instr lights (PLT)	X			
Flt instr lights (COPLT)	X			
Nonflight instr lights	X			
Console lights, upper	X			
Console lights, lower	X			
Position lights	X			
Formation lights	X			
Anticollision lights	X			
NVG lighting	X			

3.2.5 Battery life evaluation

Battery Life Evaluation
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: 325M
Serial number: 9205001
Military item number: None

Options installed: None

Manufacturer battery life specification: Up to 1 hour operation.

Overall performance: Pass

Performance: 1 hour and 4 minutes average battery life on fully charged battery.

Comments: None

3.2.6 Electrical safety test

Electrical Safety Test
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 5 Jun 92

Performance:

Grounding conductor resistance (milliohms): NA

Leakage current - Case to ground (microamperes):

unit off, grounded, normal polarity	NA
unit off, ungrounded, normal polarity	NA
unit off, ungrounded, reverse polarity	NA
unit on, grounded, normal polarity	NA
unit on, ungrounded, normal polarity	NA
unit on, ungrounded, reverse polarity	NA

MAXIMUM LIMITS:

ground resistance (milliohms):	150
current (microamperes)	
current (grounded, type A unit):	10
current (ungrounded, type A unit):	100
current (grounded, type B unit):	50
current (ungrounded, type B unit):	500

Comments on item setup or checks: None

Comments on test run (including interruptions): There were no patient leads or power lines available for testing.

Comments on other data: None

3.2.7 Human factors evaluation

Human Factors Evaluation
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 5 Jun 92

Item configuration during test: Item prepared for operation.

Checklist for HFE

	RESULTS
VISUAL DISPLAYS:	Satisfactory
display type, format, content	
location of displays	
indicator lights	
scalar displays	
color coding	
legends and labels	
cathode ray tubes	
counters	
flags, go-no-go, center-null indicators	
Comments: None	
CONTROLS:	Satisfactory
location	
characteristics of controls	
labeling	
control - display relationships	
Comments: None	

TIME REQUIRED TO PREPARE FOR OPERATION (list in comment)

Comments: approximately 2 minutes.

MAINTAINABILITY: Satisfactory

component location
component characteristics
rests and stat.'s
covers, cases, access doors
handles
lubrication
component mounting
cord storage provisions
external accessibility
internal accessibility
list special tools required
list realistic inspection requirements
list realistic inspection intervals

Comments: Operational checks should be performed as used and maintenance inspection every 6 months.

CONDUCTORS: Satisfactory

binding and securing
length
protection
routing
conductor coding
fabrication
connectors

Comments: None

FASTENERS: Satisfactory

access through inspection panel covers
enclosure fasteners
device mounting bolts and fasteners

Comments: None

TEST POINTS: Satisfactory

general
location and mounting
test point labeling and coding

Comments: None

TEST EQUIPMENT: Satisfactory

general
equipment self-test
indicators (list in comments)
controls
positive indication of proper operation

Comments: The unit is equipped with a vacuum gauge.

FUSES AND CIRCUIT BREAKERS: Satisfactory

external accessibility
easy replacement or reset by operator

Comments: None

LABELS AND CODING: Satisfactory

placed above controls and displays
near or on the items they identify
not obscured by other equipment components
describe the function of the items they identify
readable from normal operating distance
conspicuous placards adjacent to hazardous items

Comments: None

SAFETY: Satisfactory

manual
materials
fire and explosive protection
operator protection from mechanical hazards
patient protection from mechanical hazards
electrical safety (operator and patient)

Comments: None

3.2.8 Altitude test

Altitude Test
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 29 May 92

Item configuration during test: Item sitting on chamber floor.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	77°F
Humidity	66% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check:
Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None

IN-TEST DATA

Time of test start: 0945

POSTTEST DATA

Posttest performance check (complete check of item and accessories):

Time of test end : 1045

Item functional (based on performance test criteria): Yes

Deviation from pretest : None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: None

3.2.9 Vibration test

Vibration Test
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 25 May 92

Item configuration during test: Item strapped down on vibration table fixture; battery operation.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

PRETEST DATA

Pretest performance check:
Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None

Ambient conditions

Temperature	77°F
Humidity	66% RH
Barometric pressure	1 atm

IN-TEST DATA

Data and performance checks during test:

Time at first check:
X: 1310 Y: 1415 Z: 0830

Item functional (based on performance test criteria): Yes

Deviation from pretest: None

Time at second check:
X: 1350 Y: 1505 Z: 0930

Item functional (based on performance test criteria): Yes

Deviation from pretest: None

POSTTEST DATA

Time at test end:
X: 1400 Y: 1515 Z: 0930

Posttest performance check (complete check of item and accessories):

Item functional (based on performance test criteria): Yes

Item intact: Yes

Deviation from pretest: None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: Test times for the three axes are on different days.

3.2.10 High temperature test

High Temperature Test
(Equipment Operating)
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 27 May 92

Item configuration during test: Unit was sitting on chamber floor, operating on battery power.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	26°C
Humidity	55% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check :
Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None
distance from north wall (meters)	0.56
distance from south wall (meters)	1.02
distance from east wall (meters)	1.57
distance from west wall (meters)	1.45
distance from ceiling (meters)	1.19
distance from floor (meters)	0.97

IN-TEST DATA

Time of test start: 1300

Performance checks during test:

First check:

Time: 1330
Temperature: 49°C
Humidity: 15% RH
Barometric pressure: 1 atm
Item functional (based on performance test criteria): Yes, all ok
Deviation from pretest: None

Second check:

Time: 1400
Temperature: 49°C
Humidity: 15% RH
Barometric pressure: 1 atm
Item functional (based on performance test criteria): Yes, all ok
Deviation from pretest: None

Third check:

Time: 1430
Temperature: 49°C
Humidity: 15% RH
Barometric pressure: 1 atm
Item functional (based on performance test criteria): Yes, all ok
Deviation from pretest: None

POSTTEST DATA

Posttest performance check:
(complete check of item and accessories)

Time of test end: 1500
Item functional (based on performance test criteria): Yes, all ok
Deviation from pretest: None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: None

3.2.11 High temperature storage test

High Temperature Test
(Equipment in Storage)
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 2 Jun 92

Item configuration during test: Sitting on chamber floor, in storage, not operating.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	27°C
Humidity	61% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check:

Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None
distance from north wall (meters)	0.56
distance from south wall (meters)	1.02
distance from east wall (meters)	1.57
distance from west wall (meters)	1.45
distance from ceiling (meters)	1.19
distance from floor (meters)	0.97

Time of test start: 0800

POSTTEST DATA

Posttest performance check:
(complete check of item and accessories)

Time of test end: 1400
Item functional (based on performance test criteria): Yes
Deviation from pretest: None

Comments on item setup or checks:
The unit was allowed to cool for 1 hour at ambient conditions before the posttest performance check was completed.

Comments on test run (including interruptions): None

Comments on other data: None

3.2.12 Low temperature test

Low Temperature Test
(Equipment Operating)
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 28 May 92

Item configuration during test: Sitting on chamber floor.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	25°C
Humidity	60% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check:
Item functional (based on performance test criteria): Pass

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None
distance from north wall (meters)	0.56
distance from south wall (meters)	1.02
distance from east wall (meters)	1.57
distance from west wall (meters)	1.45
distance from ceiling (meters)	1.19
distance from floor (meters)	0.97

Time of test start: 0845

Performance checks during test:

First check:

Time:	0915
Temperature:	-25°C
Humidity:	NA
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Second check:

Time:	0945
Temperature:	-25°C
Humidity:	NA
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Third check:

Time:	1015
Temperature:	-25°C
Humidity:	NA
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

POSTTEST DATA

Posttest performance check:
(complete check of item and accessories)

Time of test end:	1045
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: None

3.2.13 Low temperature storage test

Low Temperature Test
(Equipment in Storage)
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Impact Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 1 Jun 92

Item configuration during test: Sitting on chamber floor, not operating, in storage.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	27°C
Humidity	49% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check:

Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None
distance from north wall (meters)	0.56
distance from south wall (meters)	1.02
distance from east wall (meters)	1.57
distance from west wall (meters)	1.45
distance from ceiling (meters)	1.19
distance from floor (meters)	0.97

Time of test start: 0900
Midtest time: 1200
Midtest temperature: -46°C

POSTTEST DATA

Posttest performance check:
(complete check of item and accessories)

Time of test end: 1500
Item functional (based on performance test criteria): Yes
Deviation from pretest: None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: None

3.2.14 Humidity test

Humidity Test
Report Form

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: Model 325M
Serial number: 9205001
Military item number: None

Options installed: None

Date of test: 29 May 92

Item configuration during test: The unit was sitting on the chamber floor, operating on ac power.

Performance test criteria: Pump must suction 250 mL of water in 4 seconds or less.

Ambient conditions outside chamber:

Temperature	25°C
Humidity	61% RH
Barometric pressure	1 atm

PRETEST DATA

Pretest performance check:
Item functional (based on performance test criteria): Yes

Installation of item in test facility:

list connections to power	None
list connections to simulators	None
list connections to dummy loads	None
list unconnected terminals	None
distance from north wall (meters)	0.56
distance from south wall (meters)	1.02
distance from east wall (meters)	1.57
distance from west wall (meters)	1.45
distance from ceiling (meters)	1.19
distance from floor (meters)	0.97

IN-TEST DATA

Time of test start: 1100

Performance checks during test:

First check:

Time:	1145
Temperature:	29.5°C
Humidity:	95% RH
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Second check:

Time:	1230
Temperature:	29.5°C
Humidity:	95% RH
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Third check:

Time:	1315
Temperature:	29.5°C
Humidity:	95% RH
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Fourth check:

Time:	1400
Temperature:	29.5°C
Humidity:	95% RH
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

Fifth check:

Time:	1415
Temperature:	29.5°C
Humidity:	95% RH
Barometric pressure:	1 atm
Item functional (based on performance test criteria):	Yes
Deviation from pretest:	None

POSTTEST DATA

Posttest performance check:
(complete check of item and accessories)
Time of test end: 1500
Item functional (based on performance test criteria): Yes
Deviation from pretest: None

Comments on item setup or checks: None

Comments on test run (including interruptions): None

Comments on other data: None

3.2.15 Electromagnetic characteristics test

****

Electromagnetic Characteristics Testing
Evaluation of Performance
****

T & E Item Number: Date: 28 May 92

Nomenclature: Suction Pump
Manufacturer: Impact Medical Corp.
Model number: 325M
Serial number: 9205001
Military item number: NA

****

Conducted Emissions Tests

CE01 Testing configuration(s): NA
Performance (pass/fail): NA

Comments: No dc conductors

CE02 Testing configuration(s): NA
Performance (pass/fail): NA

Comments: No external power leads.

CE04 Testing configuration(s): NA
Performance (pass/fail): NA

Comments: No external power leads.

Conducted Susceptibility Tests

CS02 Testing configuration(s): NA
Performance (pass/fail): NA

Comments: No external power leads.

CS06 Testing configuration(s): NA
Performance (pass/fail): NA

Comments: No external power leads.

Radiated Emissions Tests

RE02 Testing configuration(s): Operating on wooden test stand in the EMC chamber, battery power.
Performance (pass/fail): Fail

Comments: BB failure 5.5 dB over specifications at 10.75 MHz.

Radiated Susceptibility Tests

RS03 Testing configuration(s): Operating on the wooden test stand in the EMC chamber.
Performance (pass/fail): Pass

Comments: None

3.3 CRITERIA, SIGNIFICANT PROBLEMS, AND SUGGESTED IMPROVEMENTS

3.3.1 Criteria

Item No.	Criteria (source)	Remarks	Applicable subparagraph
1	The physical inventory is conducted solely for investigation and documentation.	NA	2.1.2.1
2	The Impact Model 325M will display consistent and accurate performance.	met	2.1.2.2
3	Verify manufacturer's specified full power internal battery life expectancy of 1 hour.	met	2.2.2
4	The Impact Model 325M will meet the limits established in NFPA 99 for electrical safety of medical equipment.	met	2.3.2
5	The Impact Model 325M will be rated satisfactory in all major categories of the evaluation. These include: Visual displays, controls, maintainability, conductors, fasteners, test points, test equipment, fuses and circuit breakers, labels and coding, and safety.	met	2.4.2
6	The Impact Model 325M will demonstrate proper operation while exposed to an altitude equivalency of 15,000 feet above sea level.	met	2.5.2
7	The Impact Model 325M will remain operational while exposed to vibrational stresses.	met	2.6.2
8	The Impact Model 325M will remain operational during the high temperature operation check.	met	2.7.2.1

9	The Impact Model 325M will remain operational after the high temperature storage.	met	2.7.2.2
10	The Impact Model 325M will remain operational during the low temperature operation check.	met	2.8.2.1
11	The Impact Model 325M will remain operational after the low temperature storage.	met	2.8.2.2
12	The Impact Model 325M will remain operational while exposed to a high humidity.	met	2.9.2
13	The Impact Model 325M will not produce emissions in excess of the limits set forth in MIL-STD-461A Notice 4, paragraph 6.13.	partially met	2.10.2.1
14	The Impact Model 325M will not malfunction when it is subjected to radiated fields as specified in MIL-STD-461A, Notice 4, paragraph 6.20.	met	2.10.2.2
15	The flight surgeon will be able to operate the Impact Model 325M without physical or functional restrictions aboard the aircraft.	met	2.11.2.1
16	The Impact Model 325M will not radiate EMI to disrupt or interfere with the other equipment or systems aboard the aircraft.	met	2.12.2.2
17	The aircraft will not radiate EMI to disrupt or interfere with the Impact Model 325M.	met	2.12.2.3

3.3.2 Significant problems which require corrective action

None

3.3.3 Suggested improvements

None

3.4 REFERENCES

3.4.1 Department of Defense. 1971. EMI characteristics, requirements for equipment. Washington, DC. MIL-STD-461A, Notice 4. February.

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3.4.3 Department of Defense. 1983. Environmental test methods and engineering guidelines. Washington, DC. MIL-STD-810D. July.

3.4.4 Department of the Army. 1987. Maintenance management procedures for medical equipment. Washington, DC. TB 38-750-2. April.

3.4.5 Underwriters Laboratory's, Inc. 1978. Standard for safety, medical and dental equipment. Chicago, Illinois. UL-544.

3.4.6 Department of Defense. 1989. Human engineering design criteria for military systems, equipment, and facilities. Washington, DC. MIL-STD-1472D. March.

3.4.7 Association for the Advancement of Medical Instruments. 1988. Human factors engineering guidelines and preferred practices for the design of medical devices. Arlington, Virginia. AAMI-HE-1988. February.

3.4.8 National Fire Protection Association. 1987. Standard for health care facilities. Quincy, Massachusetts. NFPA 99. February.

3.4.9 Department of the Army. 1982. Environmental protection and enhancement. Washington, DC. AR 200-1. June.

3.5 ABBREVIATIONS

ac	alternate current
AVSCOM	Army Aviation Systems Command
AWR	airworthiness release
BB	broadband
CAAF	Cairns Army Airfield
dc	direct current
EMC	electromagnetic compatibility
EMI	electromagnetic interference
fpm	feet per minute
GFE	government furnished equipment
Gpk	gravity, peak
G(rms)	gravity (root mean square)
Hz	hertz
IAW	in accordance with
ITOP	in-flight test operating procedure
IV	intravenous
kHz	kilohertz
LCD	liquid crystal display
LED	light emitting diode
LISN	line impedance stabilization network
MEDEVAC	medical evacuation
MHz	megahertz
MIL-STD	military standard
mL	milliliter
mm	millimeter
mmHg	millimeters of Mercury
MSL	mean sea level
NFPA	National Fire Prevention Association
NB	narrowband
NBC	nuclear, biological and chemical
NOE	nap-of-the-earth
NVG	night vision goggle
RF	radio frequency
RFI	radio frequency interference
RH	relative humidity

TB	technical bulletin
TFT	technical feasibility testing
T & E	test and evaluation
UES	Universal Energy Systems, Inc.
USAARL	U.S. Army Aeromedical Research Laboratory
V/m	volts per meter

3.6 LIST OF MANUFACTURERS

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3.6.3 Tenney Engineering, Inc.
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3.6.4 Unholtz-Dickey Corporation
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3.6.5 Solar Electronics Company
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